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**A Survey of the Reaction Rate
Constants for the Thermal
Dissociation and Recombination of
Nitrogen and Oxygen**

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A SURVEY OF THE REACTION RATE CONSTANTS FOR
THE THERMAL DISSOCIATION AND RECOMBINATION OF
NITROGEN AND OXYGEN

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ABSTRACT

The objective of the present report is to survey the various values of forward and backward reaction rate constants used up to date by investigators in the field of high temperature ($T > 2000^{\circ}\text{K}$) gas reactions involving nitrogen and oxygen only. By comparing the various published rate constants, which were either estimated by the reference authors or adopted from the works of others, it is examined how closely these various values of reaction rate constants are reproduced. The objective is to find those values that correlate well so that they can be used for the studies of hypersonic flow and supersonic combustion with a reasonable confidence. The relatively good agreement among these various values is observed for the temperature lower than $10,000^{\circ}\text{K}$.

INTRODUCTION

The report lists various values of forward and backward reaction rate constants used up to date by investigators in the field of high temperature ($T \gtrsim 2000^{\circ}\text{K}$) gas reactions involving nitrogen and oxygen. The reactions considered in this paper are:



and



A few of the reported values were determined experimentally. Some of the reported values were obtained by assuming a particular set of reactions and then modifying the rate constants until a satisfactory match could be made between predicted and experimentally determined concentration histories.

All reaction rate constants can be computed from:

$$k = aT^b e^{-c/T} \quad (3)$$

where T is the absolute temperature in degrees Kelvin so that the units of k are $(\text{cm}^3/\text{mole})^n \text{sec}^{-1}$. The values of a , b , and c are tabulated (Tables 1(a) to 2(c)) with the forward reaction rate constant, k_f , on the left and the backward reaction rate constant, k_b , on the right. Above each set of a , b , and c values is the value of exponent n for that reaction. The applicable or used temperature ranges reported by the various investigators, and the reference number of the publication source of each listed value are also tabulated. All the listed values are plotted for easier visual comparison. Specifically, there are five sets of four figures. Each set corresponds to one of the above mentioned reactions and contains plots of $\log_{10} k_f$ vs. T , $\log_{10} k_b$ vs. T , K vs. T , and $\log_{10} K$ vs. T . Here, K are the equilibrium constants based on concentrations. They were computed as:

$$K = k_f/k_b \quad (4)$$

Notice that in ref. 29:

$$K = \exp(A_1 + A_2 Z + A_3 Z^2 + A_4 Z^3 + A_5 Z^4) \quad (5)$$

where

$$Z = 10,000/T \quad (6)$$

One value for K was used for the reactions 1(a)-1(b) and another value for K was used for the reactions 2(a)-2(c). The corresponding values of the coefficients (ref. 29) are:

	A_1	A_2	A_3	A_4	A_5
1(a) or 1(b)	3.898	-12.611	0.683	-0.118	0.006
2(a), 2(b), or 2(c)	1.335	-4.127	-0.616	0.093	-0.005

The values of k_f for these reactions were given in ref. 29. We computed the corresponding values of k_b from the formulas (4), (5), and (6) with the help of the above table.

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CONCLUSION

An attempt has been made to correlate the published values for the reaction rate constants involving nitrogen and oxygen only. From the tabulated and plotted data it is possible to judge the validity of the individual rate constants in the temperature ranges considered here. It is obvious that there is a discrepancy of the published data for higher temperatures. Reliability of the published values for the backward reaction rates is highly questionable, thus, warranting a more accurate experimental and analytical verification of these parameters.

ACKNOWLEDGMENTS

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BIBLIOGRAPHY

1. Zel'dovich, Ya. B. and Rayzer, Yu. P., "Physics of Shock Waves and High Temperature Hydrodynamics Phenomena," FTD-MT-64-514 machine translation, 1963; also, Physics of Shock Waves and High-Temperature Hydrodynamic Phenomena, Vol. 1, Academic Press, 1966.
2. Gibson, W. E. and Marrone, P. V., "A Similitude for Non-Equilibrium Phenomena in Hypersonic Flight," The High Temperature Aspects of Hypersonic Flow, (AGARDograph 68), The Mac Millan Co. (1964), Chapter 11, pp. 217-254; also, Cornell Aeronautical Lab. Report AF-1500-A-1, July 1962.
4. Magnus, D. E., and Schechter, H. S., "Analysis of Error Growth and Stability for the Numerical Integration of the Equations of Chemical Kinetics," General Applied Science Labs., Inc., Report NASA CR-66371, 1964.
5. Gavril, B. D., "Generalized One-Dimensional Chemically Reacting Flows with Molecular Vibration Relaxation," General Applied Science Labs., Inc., Technical Report GASL-TR-426, January 29, 1964.
6. Duffy, R. E., "Experimental Studies of Non-Equilibrium Expanding Flows," Ph.D. thesis, Department of Aeronautical Engineering and Astronautics, Rensselaer Polytechnic Institute, Troy, New York, April 1965.
7. Bortner, M. H. and Galbraith, H. J., "Study of the Accuracy of Rate Constants for Deionization Calculations. Final Report," DASA-1667, May 31, 1965; also Bortner, M. H., "Chemical Kinetics in a Reentry Flow Field," General Electric Space Sciences Lab., Report GE-R63 SD63, August 1963.
8. Sherman, M. P., "Radiation-Coupled Chemical Nonequilibrium Normal Shock Waves," General Electric Space Sciences Lab., Report GE-R66 SD17, March 1966.
9. Bortner, M. H., "Suggested Standard Chemical Kinetics for Flow Field Calculations - A Concensus Opinion," AMRC Meeting Proceedings Volume 14, Part 1, April 19, 1966, pp. 569-581.
10. Hanson, R. K., and Watson, R., "Effects of Dissociation Rate Magnitudes and Relative Collision Efficiencies on Relaxation Profiles in Diatomic Gases," AIAA Journal 4, No. 4, April 1966, pp. 749-751.
11. Sarkos, C., "Pure Air Wake Flow Field Computations," General Electric Space Sciences Lab., Report GE-R66 SD338, November 23, 1966.
12. Heicklen, J. P., "Gas-Phase Chemistry of Reentry," AIAA Journal 5, No. 1, January 1967, pp. 4-15; also, Aerospace Corporation, Report TDR-669 (6250-40)-7, May 1966.
13. Wen, K. S., Chen, T., Hayami, R. A., and Primich, R. I., "An Eddy Diffusion Model for Predicting Rapid Wake Ionization Decay Behind Hypersonic Cones," Paper No. 67-21, AIAA 5th Aerospace Sciences Meeting, January 23-27, 1967; also, Wen, K. S., and Chen, T., "Axisymmetric Viscous Wake Analysis for a Hypersonic Reentry Body," General Motors Defense Research Labs. Technical Report TR-66-12a, April 1966.
14. Zonars, D., "Nonequilibrium Regime of Airflows in Contoured Nozzles: Theory and Experiments," AIAA Journal 5, No. 1, January 1967, pp. 57-63.
15. Bauer, E., "The Excitation of Electronic and Other Degrees of Freedom in a Hypersonic Shock Wave in Air," Institute for Defense Analysis Research Paper, P-311, IDA-HQ-67-5724, AD-650950, January 1967.
16. McMenamin, D., and O'Brien, M., "The Finite Difference Solution of Multicomponent Nonequilibrium Steady Inviscid Streamtube Flows Using a Novel Stepping Technique. Part I. Analysis and Applications," General Electric Aerospace Physics Lab. Report 67 SD 241, April 1967.
17. Nerem, R. M., Carlson, L. A., and Hartsel, J. E., "Chemical Relaxation Phenomena Behind Normal Shock Waves in a Dissociated Freestream," AIAA 4th Aerospace Sciences Meeting, June 27-29, 1966.
18. Clyne, M. A., and Stedman, D. H., "Rate of Recombination of Nitrogen Atoms," J. Phys. Chem. 71, No. 9, August 1967, pp. 3071-3073.
19. DASA Reaction Rate Handbook, DASA Information and Analysis Center, DASA 1948, October 1967.

20. Curtis, J. T., and Strom, C. R., "Computations of the Nonequilibrium Flow of a Viscous Radiating Fluid about a Blunt Axisymmetric Body, Volume 1: Equations and Results," Cornell Aeronautical Lab. Inc., AFFDL-TR-67-40, Vol. 1, AD 660104, June 1967.
21. Wilson, J., "A Measurement of the Recombination Rate of Oxygen," The High Temperature Aspects of Hypersonic Flow, (AGARDograph 68), The Mac Millan Company, (1964), Chapter 23, pp. 471-483.
22. Lun'kin, Yu. P., and Popov, F. D., "Influence of Vibration-Dissociation Relaxation on Supersonic Flow Around Blunt Bodies," Soviet Phys.-Technical Phys. 11, No. 4, October 1966, pp. 491-497.
23. Conor, L. N., Jr., "Calculation of The Centered One-Dimensional Unsteady Expansion of a Reacting Gas Mixture Subject to Vibrational and Chemical Nonequilibrium," NASA TN D-3851, February 1967.
24. De Jarnette, F. R., "Two Different Interpretations of Measured Dissociation-Rate Constants and Their Effects on Coupled Vibrational-Dissociational Flows of Oxygen Over a Wedge," NASA TN D-4028, July 1967.
25. Bauer, T., Ciokowski, S., Weeker, M. S., and Attwood, D., "Upper Atmospheric Phenomenology: A Comparative Study of Effects Due to Hypersonic Reentry and High-Altitude Nuclear Events," General Applied Science Labs., Final Report, DASA 1834, August 1966.
26. Appleton, J. P., Steinberg, M., and Liquornik, D. J., "Shock-Tube Study of Nitrogen Dissociation using Vacuum-Ultraviolet Light Absorption," J. Chem. Phys. 48, No. 2, January 15, 1968, pp. 599-608.
27. Rakich, J. V., Bailey, H. E., and Park, C., "Computation of Nonequilibrium Three-Dimensional Inviscid Flow over Blunt-Nosed Bodies Flying at Supersonic Speeds," AIAA Paper No. 75-835, AIAA 8th Fluid and Plasma Dynamics Conference, Hartford, CT, June 16-18, 1975.
28. Song, D. J., and Lewis, C. H., "Hypersonic Finite Rate Chemically Reacting Viscous Flows over An Ablating Carbon Surface," Presented as Paper 84-1731 at The AIAA 19th Thermophysics Conference, Snowmass, CO, June 25-28, 1984.
29. Park, C., "Convergence of Computation of Chemically Reacting Flows," Presented as Paper 85-0247 at the AIAA 23rd Aerospace Conference Meeting, Reno, NV, January 14-17, 1985.
30. Hall, J. G., Eschenroeder, A. Q., and Marrone, P. V., "Blunt-Nose Inviscid Airflows with Coupled Nonequilibrium Processes," Cornell Aeronautical Lab., Inc., June 18, 1962.
31. Kang, S. W. and Dunn, M. G., "Theoretical and Measured Electron-Density Distributions for the RAM Vehicle at High-Altitudes," AIAA Paper 72-689, June 1972.

Table 1(a). Reaction Rate Constants

ref	k_i (n=1)			k_b (n=2)			Applicable Temp.
	a	b	c	a	b	c	
1				2.2×10^{14}	0.5	0	
2				1.5×10^{20}	1.5	0	
3	3.0×10^{21}	-1.5	113,260				
4	2.8×10^{19}	-1.0	113,200	2.0×10^{18}	-1.0	0	7,400-23,000
5				2.8×10^{16}	-0.5	0	
6	3.0×10^{21}	-1.5	113,220				
7	3.8×10^{18}	-1.0	113,200	2.0×10^{18}	-1.0	0	1,350-7,000
8				2.0×10^{18}	-1.0	0	
9	4.8×10^{17}	-0.5	113,100	2.6×10^{16}	-0.5	0	high-temp.
10				1.78×10^{17}	-0.5	0	
11	4.8×10^{17}	-0.5	113,000				0-6,000
12	7.1×10^{19}	-1.0	113,225				6,000-10,000
13				1.5×10^{20}	-1.5	0	
13	3.0×10^{21}	-1.5	132,600				300-3,100
14	4.75×10^{17}	-0.5	113,171				300-4,500
15	4.8×10^{17}	-1.5	113,100	2.6×10^{18}	-0.5	0	4,120-5,725
16	1.0×10^{22}	-1.5	113,232	9.55×10^{20}	-1.57	0	
17				1.5×10^{20}	-1.5	0	
18				5.0×10^{14}	0	-503	7,300-10,300
19				1.4×10^{21}	-1.7	0	± 151
20				1.5×10^{20}	-1.5	0	
26	3.7×10^{21} ± 1.0	-1.6	113,200	2.2×10^{20} ± 0.6	-1.6	0	8,000-15,000
27				3.0×10^{14}	0	-500	3,000-10,000
28	1.92×10^{17}	-0.5	113,100	1.09×10^{16}	-0.5	0	1,000-3,000
29	3.7×10^{21}	-1.6	113,200				500-50,000
30	3.0×10^{21}	-1.5	113,260	1.67×10^{20}	-1.5	810	
31	4.7×10^{17}	-0.5	113,200				

Table 1(b). Reaction Rate Constants

ref	k_i (n=1)			k_b (n=2)			Applicable Temp.
	a	b	c	a	b	c	
2				7.5×10^{20}	-1.5	0	
3	1.5×10^{20}	-1.5	113,260				
4	1.3×10^{20}	-1.0	113,200	7.0×10^{18}	-1.0	0	7,400-23,000
5				2.4×10^{21}	-1.5	0	
6	1.5×10^{22}	-1.5	113,220				1,350-7,000
7	1.3×10^{21}	-1.0	113,200	7.0×10^{19}	-1.0	0	
8				7.0×10^{18}	-1.0	0	10,000
9	4.2×10^{22}	-1.5	113,100	2.3×10^{21}	-1.5	0	high-temp.
11	4.2×10^{22}	-1.5	113,000				0-6,000
12				5.0×10^{15}	0	0	700-6,800
12	7.1×10^{19}	-1.0	113,225				6,000-10,000
13				7.5×10^{20}	-1.5	0	300-3,100
13	1.5×10^{22}	-1.5	132,600				300-4,500
14	4.1×10^{22}	-1.5	113,171				4,120-5,725
15	4.2×10^{22}	-1.5	113,100	2.3×10^{21}	-1.5	0	
16	6.0×10^{21}	-1.5	113,232	5.73×10^{20}	-1.57	0	$\pm 2,000$
17				7.5×10^{20}	-1.5	0	7,300-10,300
20				7.5×10^{20}	-1.5	0	
26	1.6×10^{22} ± 0.6	-1.6	113,200	9.8×10^{20} ± 3.6	-1.6	0	8,000-15,000
27				9.0×10^{14}	0	-500	3,000-10,000
28	4.15×10^{22}	-1.5	113,100	2.32×10^{21}	-1.5	0	1,000-3,000
29	1.11×10^{22}	-1.6	113,200				500-50,000
30	1.5×10^{22}	-1.5	113,260	8.33×10^{20}	-1.5	810	
31	4.1×10^{22}	-1.5	113,200				

Table 2(a). Reaction Rate Constants

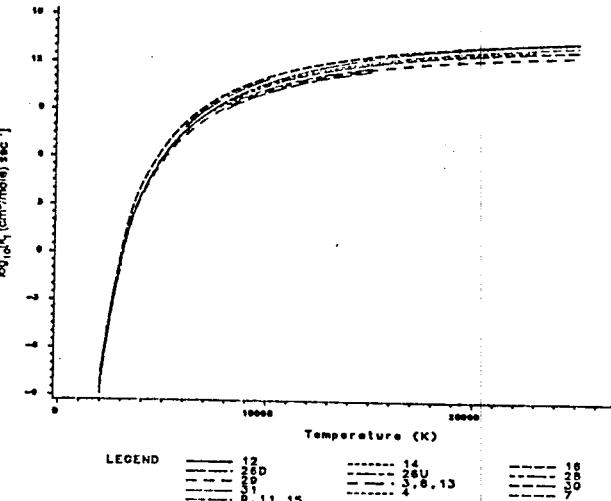
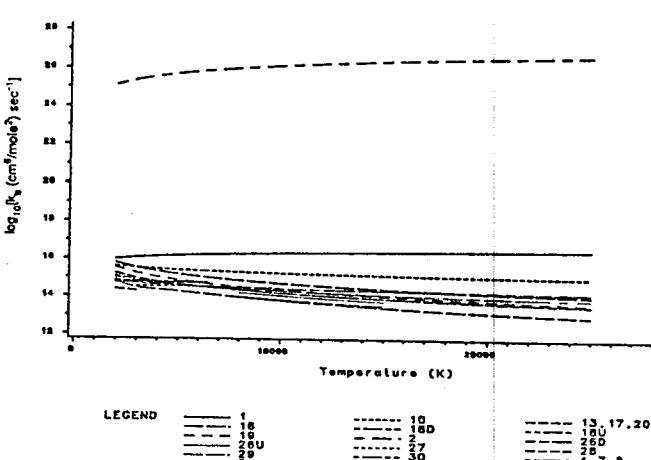
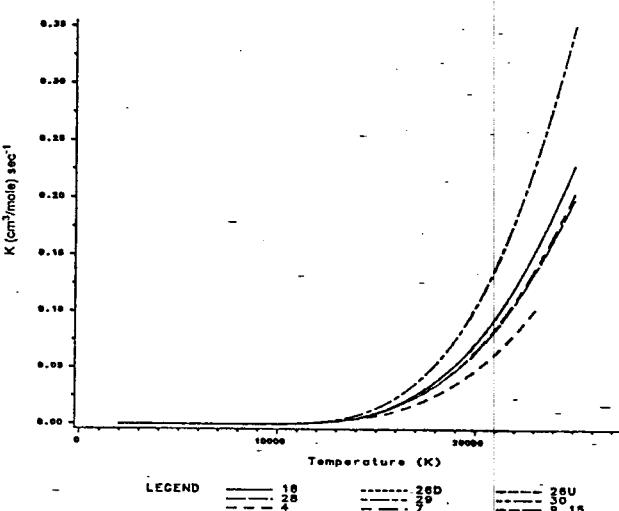
$O_2 + O_2 \rightleftharpoons O_2 + 2O$							
$k_f (n=1)$			$k_b (n=2)$			Applicable Temp.	
ref	a	b	c	a	b	c	
2	3.6×10^{21}	-1.5	59,380				
3	3.6×10^{21}	-1.5	59,380				
4	2.3×10^{19}	-1.0	59,400	1.9×10^{16}	-0.5	0	7,400-23,000
5				8.0×10^{19}	-1.5	0	
6	3.6×10^{21}	-1.5	59,360				1,350-7,000
7	2.3×10^{19}	-1.0	59,400	1.9×10^{16}	-0.5	0	
8				1.9×10^{16}	-0.5	0	10,000
9	3.3×10^{19}	-1.0	59,400	2.7×10^{16}	-0.5	0	high-temp.
11	3.3×10^{19}	-1.0	59,300				0-6,000
12	1.0×10^{16}	0	59,380				2,800-5,000
13	3.56×10^{21}	-1.5	59,380				300-3,100
14	3.26×10^{19}	-1.0	59,378				4,120-5,725
15	3.3×10^{19}	-1.0	59,400	2.7×10^{16}	-0.5	0	≥ 2000
16	1.3×10^{17}	-0.5	59,352	4.5×10^{15}	-0.44	0	
17	3.6×10^{21}	-1.5	59,380				
19				6.9×10^{22}	-2.5	0	2,500-5,000
20	3.56×10^{21}	-1.5	59,380		± 0.5		
21				2.5×10^{18}	-1.0	0	
22	1.06×10^{25}	-2.5	59,395				
23	3.258×10^{19}	-1.0	59,395	2.715×10^{16}	-0.5	0	
24	1.1×10^{25}	-2.5	59,380				1,500-8,000
27				1.38×10^{18}	-1.0	171.5	3,000-10,000
29	2.75×10^{19}	-1.0	59,500				500-50,000
30	3.6×10^{21}	-1.5	59,380	3.0×10^{18}	-1.0	380	
31	3.2×10^{19}	-1.0	59,500				

Table 2(b). Reaction Rate Constants

$O_2 + O \rightleftharpoons O + 2O$							
$k_f (n=1)$			$k_b (n=2)$			Applicable Temp.	
ref	a	b	c	a	b	c	
2	2.1×10^{18}	-0.5	59,380				
3	2.1×10^{18}	-0.5	59,380				
4	8.5×10^{19}	-1.0	59,400	7.1×10^{16}	-0.5	0	7,400-23,000
5				2.3×10^{20}	-1.5	0	
6	2.1×10^{18}	-0.5	59,360				1,350-7,000
7	8.5×10^{19}	-1.0	59,400	7.1×10^{16}	-0.5	0	
8				7.1×10^{16}	-0.5	0	10,000
9	9.0×10^{19}	-1.0	59,400	7.6×10^{16}	-0.5	0	high-temp.
11	9.0×10^{19}	-1.0	59,300				0-6,000
12	3.0×10^{16}	0	59,380				2,800-5,000
13	2.1×10^{18}	-0.5	59,380				300-3,100
14	9.04×10^{19}	-1.0	59,378				4,120-5,725
15	9.0×10^{19}	-1.0	59,400	7.6×10^{17}	-0.5	0	$\geq 2,000$
16	6.3×10^{17}	-0.5	59,352	2.2×10^{16}	-0.44	0	
17	2.1×10^{18}	-0.5	59,380				
20	2.1×10^{18}	-0.5	59,380				
24	1.1×10^{25}	-2.5	59,380				1,500-8,000
27				4.14×10^{18}	-1.0	171.5	3,000-10,000
29	8.25×10^{19}	-1.0	59,500				500-50,000
30	2.1×10^{18}	-0.5	59,380	1.75×10^{15}	0	380	
31	2.0×10^{19}	-1.0	59,500				

Table 2(c). Reaction Rate Constants

$O_2 + N_2 \rightleftharpoons N_2 + 2O$							
$k_f (n=1)$			$k_b (n=2)$			Applicable Temp.	
ref	a	b	c	a	b	c	
5				6.2×10^{15}	-0.5	0	
7				2.0×10^{16}	-0.5	0	
9	7.2×10^{18}	-1.0	59,400	6.2×10^{15}	-0.5	0	high-temp.
11	7.2×10^{18}	-1.0	59,300				0-6,000
14	7.23×10^{18}	-1.0	59,378				4,120-5,725
15	7.2×10^{18}	-1.0	59,400	6.2×10^{15}	-0.5	0	$\geq 2,000$
19				6.9×10^{22}	-2.5	0	2,500-5,000
23	7.24×10^{18}	-1.0	59,395	6.033×10^{15}	-0.5	0	
27				1.38×10^{18}	-1.0	171.5	3,000-10,000
29	2.75×10^{19}	-1.0	59,500				500-50,000
30	1.2×10^{21}	-1.5	59,380	1.0×10^{18}	-1.0	380	
31	7.2×10^{18}	-1.0	59,500				

Fig. 1(a)-1 Forward Reaction Rate Kf $N_2 + N_2 \rightleftharpoons N_2 + 2N$ Fig. 1(a)-2 Backward Reaction Rate Kb $N_2 + N_2 \rightleftharpoons N_2 + 2N$ Fig. 1(a)-3 Equilibrium Constants K $N_2 + N_2 \rightleftharpoons N_2 + 2N$

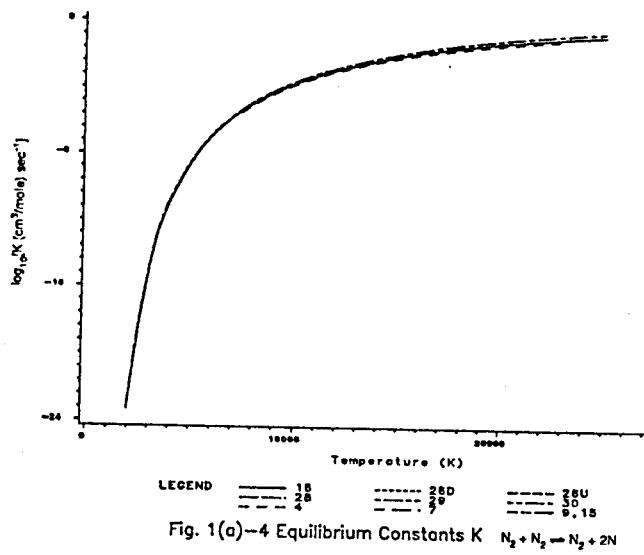


Fig. 1(a)-4 Equilibrium Constants K $N_2 + N_2 \rightleftharpoons N_2 + 2N$

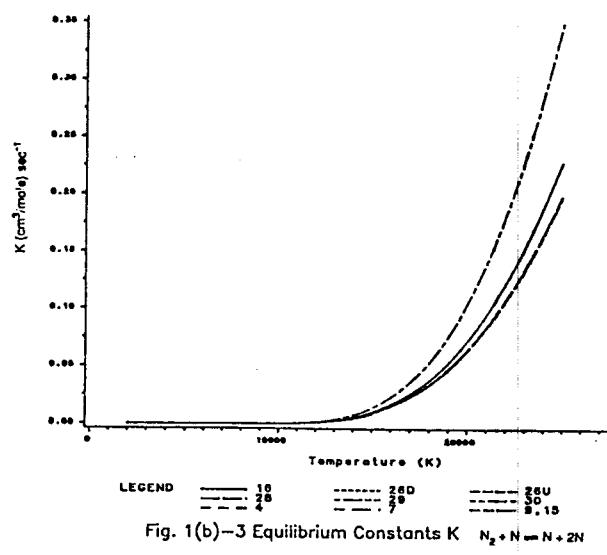


Fig. 1(b)-3 Equilibrium Constants K $N_2 + N \rightleftharpoons N + 2N$

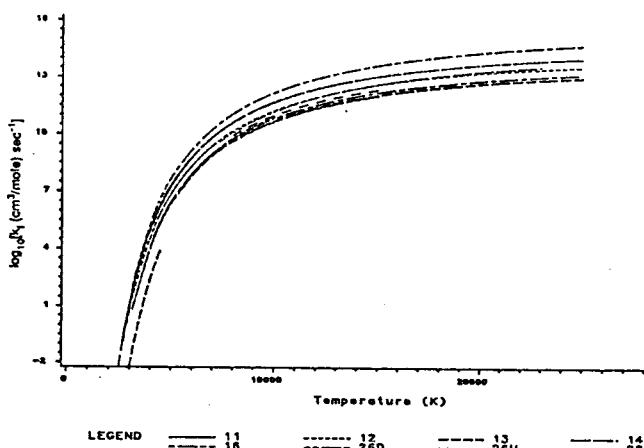


Fig. 1(b)-1 Forward Reaction Rate Kf $N_2 + N \rightleftharpoons N + 2N$

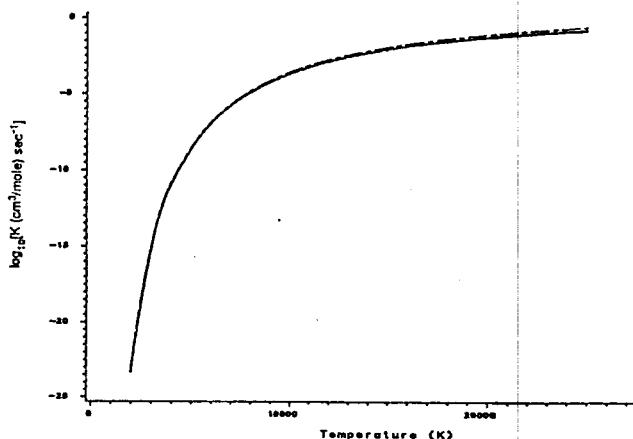


Fig. (1b)-4 Equilibrium Constants K $N_2 + N \rightleftharpoons N + 2N$

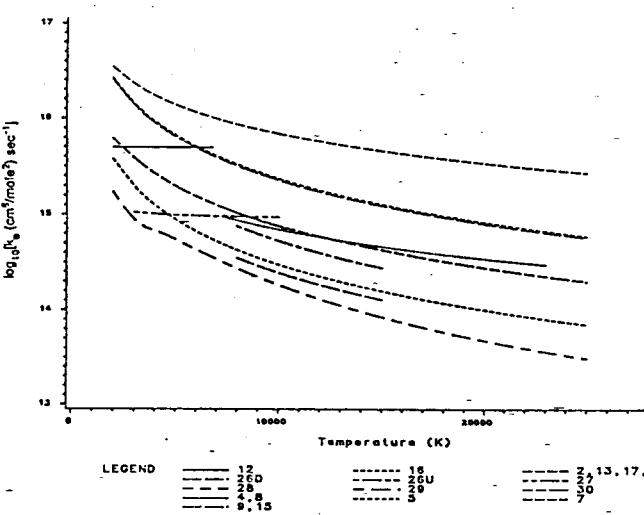


Fig. 1(b)-2 Backward Reaction Rate Kb $N_2 + N \rightleftharpoons N + 2N$

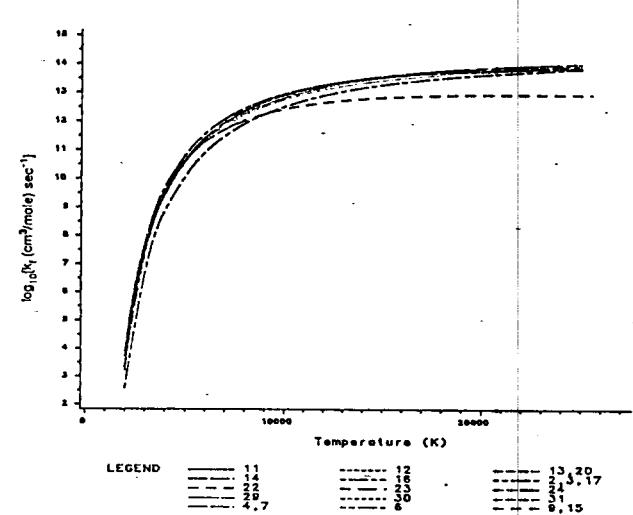
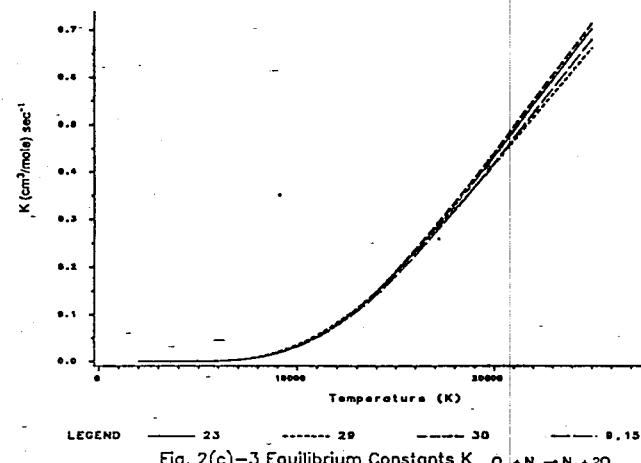
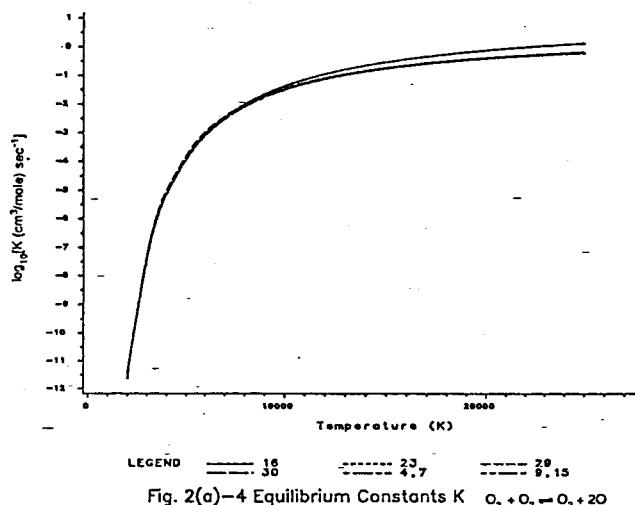
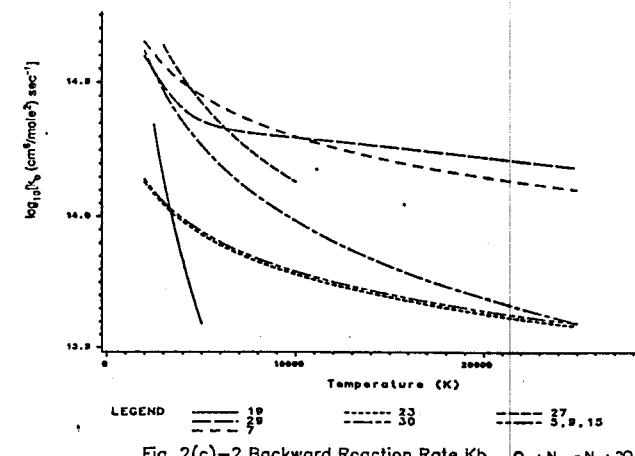
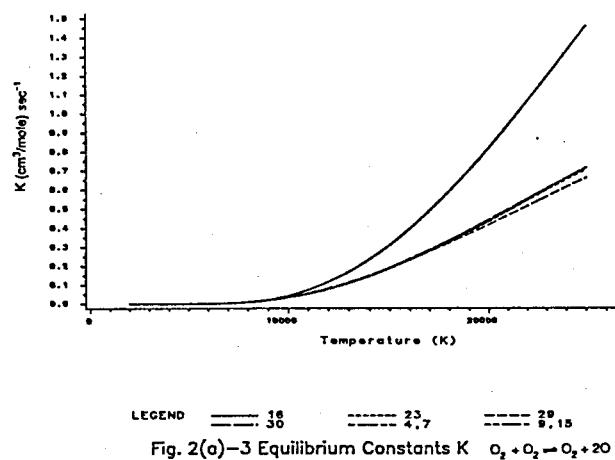
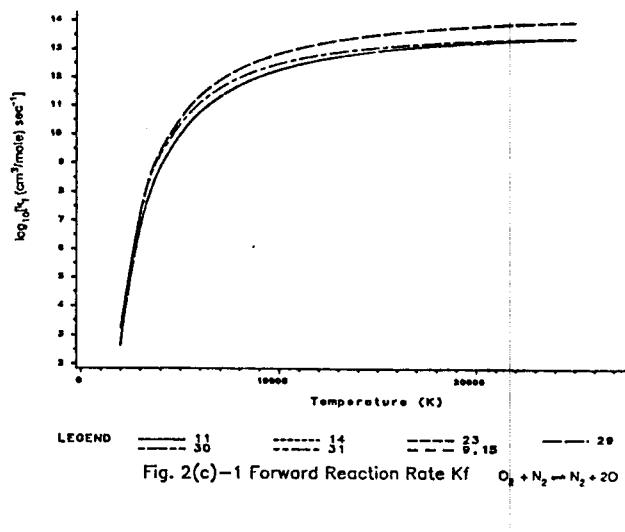
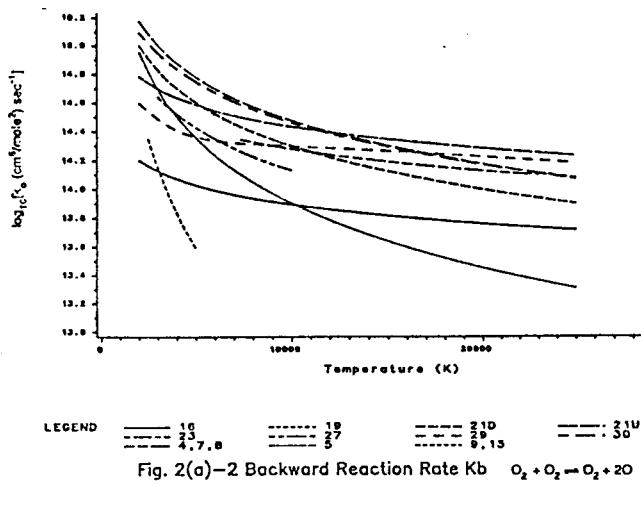


Fig. 2(a)-1 Forward Reaction Rate Kf $O_2 + O_2 \rightleftharpoons O_2 + 2O$



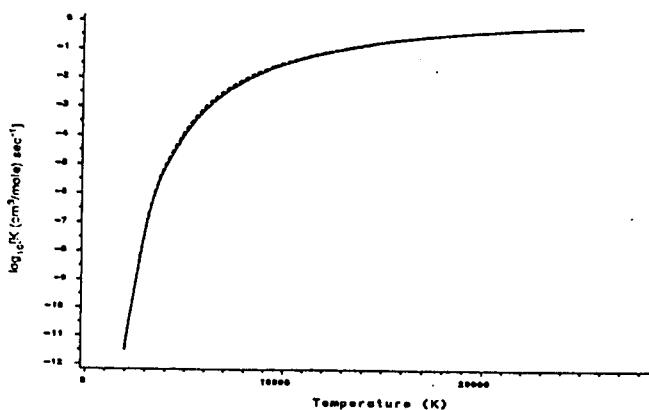


Fig. 2(c)-4 Equilibrium Constants K $O_2 + N_2 \rightleftharpoons N_2 + 2O$

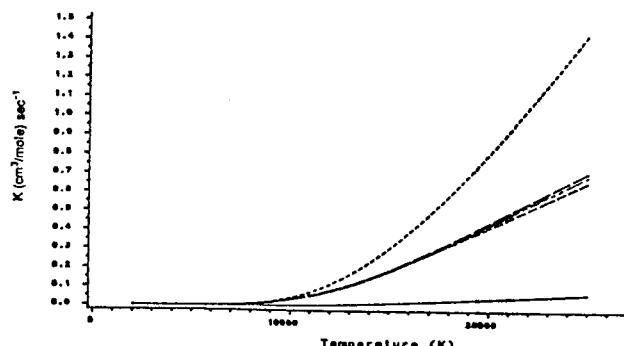


Fig. 2(b)-3 Equilibrium Constants K $O_2 + O \rightleftharpoons O + 2O$

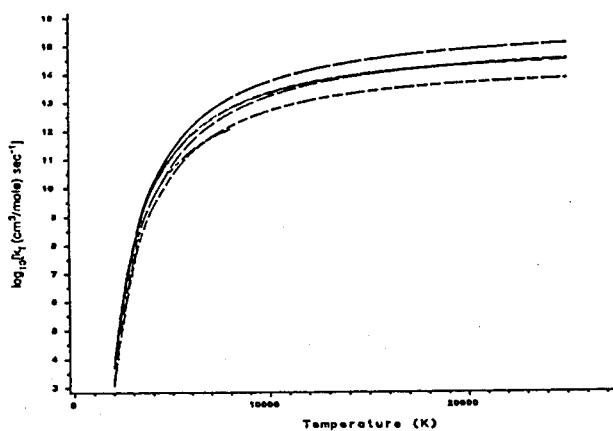


Fig. 2(b)-1 Forward Reaction Rate K_f $O_2 + O \rightleftharpoons O + 2O$

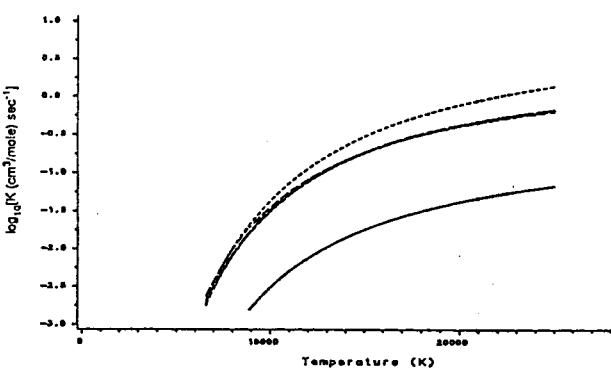


Fig. 2(b)-4 Equilibrium Constants K $O_2 + O \rightleftharpoons O + 2O$

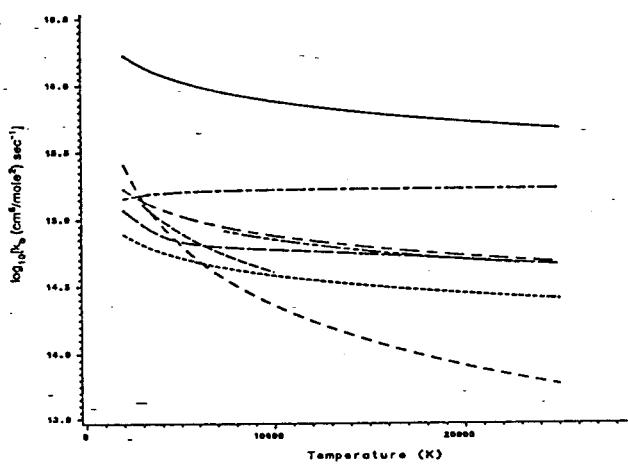


Fig. 2(b)-2 Backward Reaction Rate K_b $O_2 + O \rightleftharpoons O + 2O$